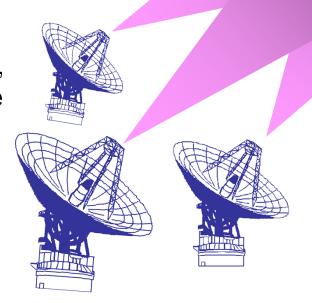


Planetary Radar Imaging with the Deep-Space Network's 34 meter Uplink Array

V. Vilnrotter, P. Tsao, D.Lee,

T. Cornish, J. Jao, M. Slade

March 11, 2011





UPLINK ARRAYING TO DEEP-SPACE IS CURRENTLY BEING DEMONSTRATED AT JPL

- MOON-BOUNCE CALIBRATION, RE-POINTING, LONG-TERM PHASE CONTROL, AND CONTINUOUS TRACKING OF REAL DEEP-SPACE PROBES HAS BEEN DEMONSTRATED
- PLANETARY RADAR IMAGING IS CURRENTLY BEING INVESTIGATED





Picture of Apollo cluster, forming transmitter part of Uplink Array (DSS-24, DSS-25, DSS-26)

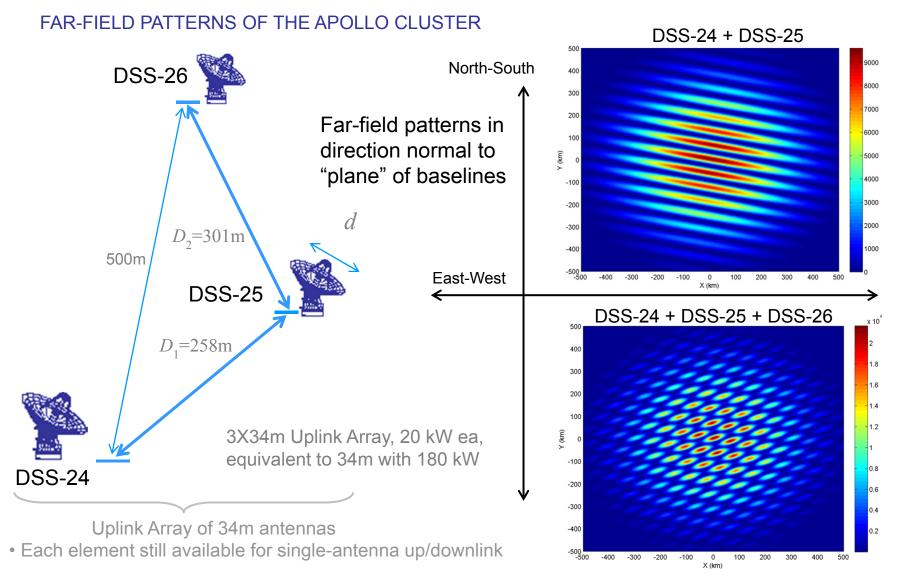
- Three 34m BWG antennas, 20 KW transmitters
- Array spans ~ 500 meters
 - Antenna null-to-null beamwidth ~ 170 mdegs
 - DSS-24 DSS-25 baseline ~ 23 mdegs
 - DSS-24 DSS-26 baseline ~ 15 mdegs

7.15 GHz Uplink Array receiver at the DSS-13 pedestal room

- Upgraded to cryogenic LNA
- X-band output to MMS
- 321.4 MHz IF to RSR (DSS-13)
- 460 MHz IF to GSSR (DSS-14)



Jet Propulsion Laboratory
California Institute of Technology

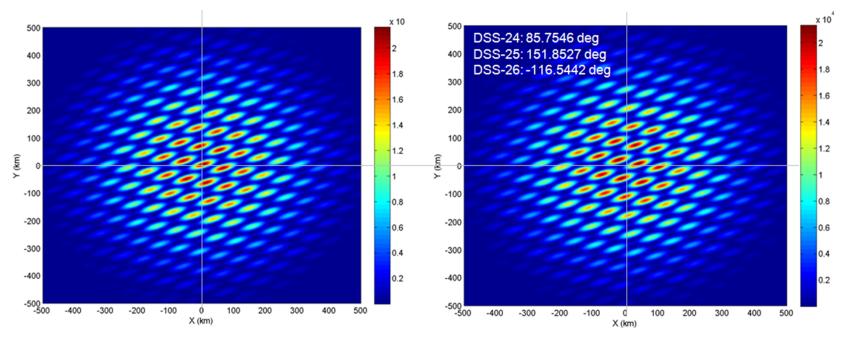




EFFECT OF RANDOM PHASE ON 3-ANTENNA ARRAY

With perfect phasing, peak of array far-field pattern maximized on-axis

With randomized phase, array far-field pattern retains basic structure, but array peak is shifted off desired direction



- Regular structure suggests simple extension of 2-antenna calibration approach:
 obtain Doppler-delay image of target; estimate phase offsets for both independent baselines
- Efficient scanning of entire primary beam possible, using a few phase-shifts



Jet Propulsion Laboratory
California Institute of Technology

SPS frequency predicts for DSS-25: f(t)

Geometry-derived phase difference, p(t):

DSS
$$n$$
, $p_n(t) = \frac{2\pi}{\lambda} \vec{\mathbf{a}}_n \cdot \vec{\mathbf{s}}(t)$

Geometry-derived frequency difference, $f_{\Lambda}(t)$:

DSS
$$n$$
, $f_{\Delta,n}(t) = \frac{2\pi}{\lambda} \vec{\mathbf{a}}_n \cdot \frac{\partial}{\partial t} \vec{\mathbf{s}}(t)$

$$f_n(t) = f(t) + f_{\Delta,n}(t)$$

Pointing-based frequency predicts derived from first principles to minimize differential phase error

DSS-26

$$\vec{\mathbf{s}}(t) = S_x(t)\vec{\mathbf{i}} + S_y(t)\vec{\mathbf{j}} + S_z(t)\vec{\mathbf{k}}$$

Standard SPS/ITT frequency

predicts for DSS-25, f(t)

DSS-26 relative position vector

$$\vec{\mathbf{a}}_{26} = a_{26,x}\vec{\mathbf{i}} + a_{26,y}\vec{\mathbf{j}} + a_{26,z}\vec{\mathbf{k}}$$

DSS-24 relative position vector

$$\vec{\mathbf{a}}_{24} = a_{24,x}\vec{\mathbf{i}} + a_{24,y}\vec{\mathbf{j}} + a_{24,z}\vec{\mathbf{k}}$$

 Precise knowledge of relative position vector coefficients is essential for accurate frequency and phase predicts

 Accurate coefficients were provided via previous VLBI solution for the DSS-24/25/26 phase centers (C. Jacobs, private communication) DSS-25. Requires

reference

k

Requires standard pointing predicts to spacecraft

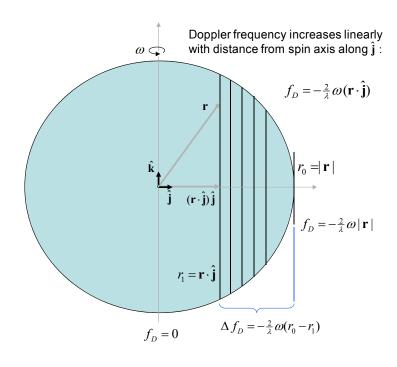
Normalized "pointing vector"

(based on pointing predicts)

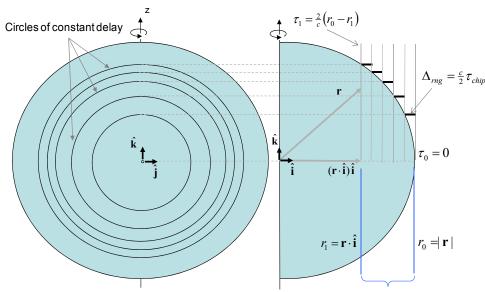
DSS-24



Rotating Sphere Model for Moon and Planets



Doppler frequency increases linearly from the center, generating constant frequency "Doppler slices" parallel to the apparent spin axis of a sphere.



Range to lunar and planetary targets known to cm accuracy

Range-depth in "range-increments" $K=(r_0-r_1)/\Delta_{mg}$ Delay-depth in "chips" $K=(\tau_1-\tau_0)/\tau_{chip}$

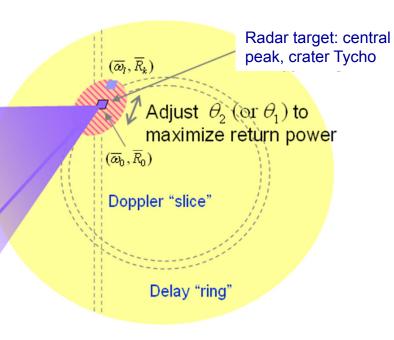
Concentric circles of increasing delay from the closest point, the sub-radar point, with definition of "range-increments" and "delay-depth" on a sphere.

Transmit-Mode Uplink Array Calibration Technique: "Moon-Bounce"

 R_1

 Maximization of far-field power provides direct calibration of antenna pairs in the uplink array

 Ranging code & Doppler compensation required to generate Doppler-delay "pixels"



 $R_2 - R_1 \qquad R_2$ $R_2 - R_1 \qquad R_2$ Antenna 2 R_0 Total instrument phase (θ_1, θ_2) characterizes each antenna 1

Receiving

antenna

- Far-field interference pattern steered electronically by adjusting local transmitter phase
- Phase difference that maximizes far-field power over "target pixel" phases up two-element array
- Known target pixel ephemeredes establish pointing direction for the array beam



CW ECHO POWER, NOISE POWER, AND SNR ESTIMATES

$$P_r = \delta_r \sigma_r \eta_{rt} P_t \left(\frac{d_r}{2R} \right)^2 = \delta_r \sigma_r \eta_{rt} P_t \left(\frac{d_r}{R} \right)^2 \qquad \delta_r = \min[1, (d_t D/R)^2]$$

$$P_n = kT_{sys}B$$

$$k = 1.38 \times 10^{-23}$$
 $d_r = 34$

$$P_t = 2 \times 10^4 \qquad \eta_{rt} = 0.5$$

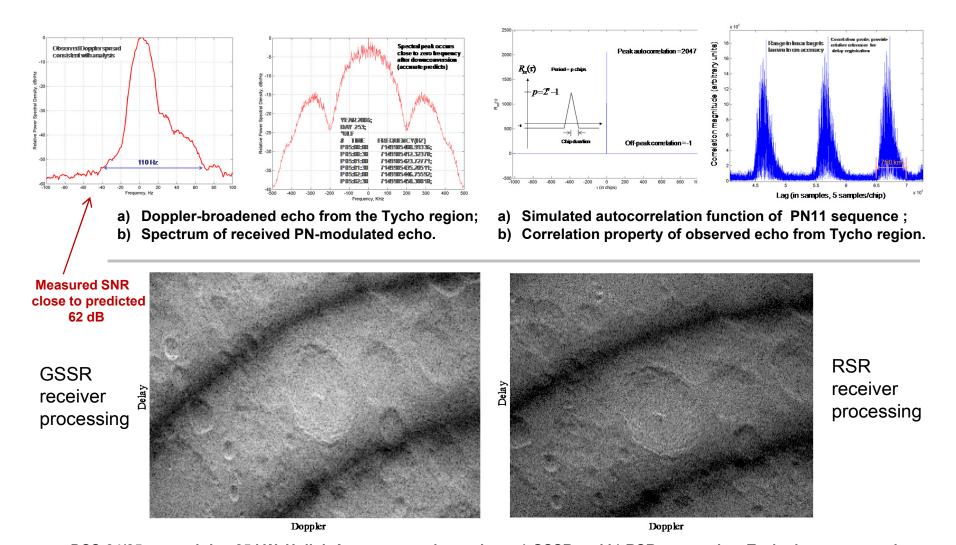
$$SNR = \frac{P_r}{P_n} = \frac{\delta_r \sigma_r \eta_{rt} P_t}{k T_{sys} B} \left(\frac{d_r^2}{R^2} \right)$$

$$\delta_r = 1, \ \sigma_r = 0.001, B = 30 \text{ Hz} \text{ and } T_{sys} = 100 \text{ k}$$
 $SNR_{Moon} \cong 1.7 \times 10^6 \quad (62 \text{ dB})$

$$\delta_r = 9 \times 10^{-4}$$
, $\sigma_r = 0.1$, $B = 300 \text{ Hz}$ and $T_{sys} = 36 \text{ k}$ $SNR_{Mercury} \cong 1$ (0 dB)

$$\delta_r = 2.25 \times 10^{-2}, \, \sigma_r = 0.1/10, \, B = 100 \,\text{Hz}, \, T_{sys} = 36 \,\text{k} \quad SNR \cong 30.8 \quad (14.9 \,\text{dB})$$

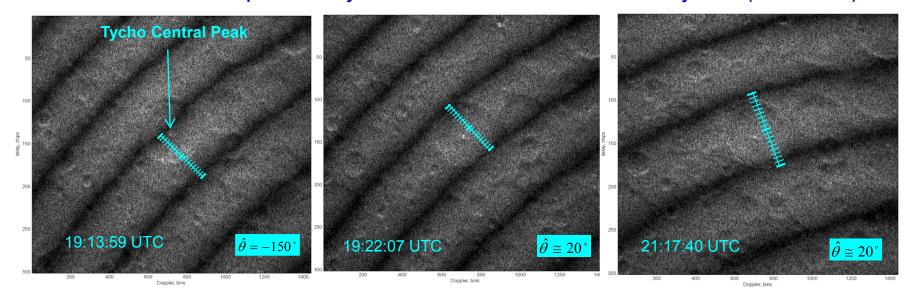
Jet Propulsion Laboratory
California Institute of Technology



DSS-24/25 transmitting 25 kW, Uplink Array cryogenic receiver: a) GSSR and b) RSR processing, Tycho image comparison.



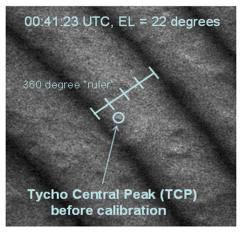
Two-Antenna Uplink Array Calibration + Far-Field Stability test (DOY-196)

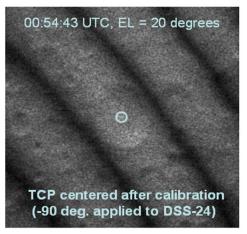


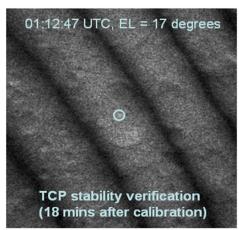
- Uplink Array calibration target (Tycho CP) initially -150 electrical degrees from fringe peak
- After rough-calibration, single correction applied to DSS-24 carrier phase
 - Phase correction placed Uplink Array fringe peak within ~ 10 electrical degrees of lunar target
- Target remained within ~ 20 degrees of peak for about 2 hours (end-of-track)
 - Real-time phase corrections were applied to DSS-24 to mitigate ground-system phase drift

Jet Propulsion Laboratory
California Institute of Technology

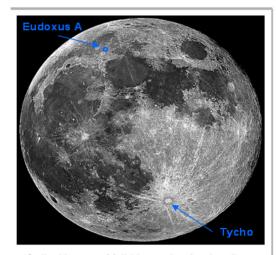
CALIBRATION-VECTOR TRANSLATION DEMO; DSS-24/25, DSS-13; DOY-257







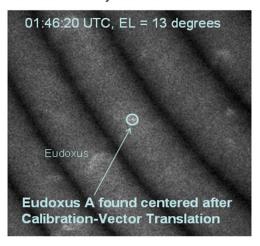
Top three Doppler-delay images: Standard calibration and stability verification via Tycho Central Peak



Optical image of full Moon, showing locations of calibration and vector-translation targets

Right: Demonstration of Calibration-Vector Translation concept via secondary target Eudoxus A, using computed phase and frequency predicts

Ground-phase was monitored at SPC-10, but no phase corrections were applied (phase difference remained within ± 5 degrees throughout the entire track)



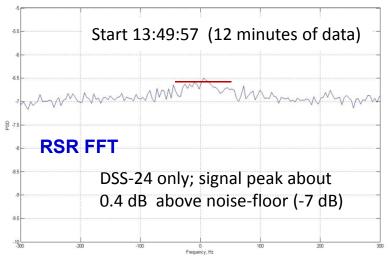


UPLINK ARRAY PLANETARY DOPPLER-DELAY IMAGING

- Radar echo power falls off as fourth power of distance: $P \propto R^{-4}$
- Mercury, Venus, and Mars can provide detectable echo power with 2-3 element Uplink Array and 34 meter receiver
- EIRP of 2-element 34m Uplink Array each with 20 kW transmitters equivalent to 70m antenna with 20 kW
- Detection of extremely weak planetary echoes requires very accurate frequency and range predicts
- First attempt to image Mercury with Uplink Array during May 1st, 2010 close approach
- First attempt to image Venus occurred on October 24th, 2010



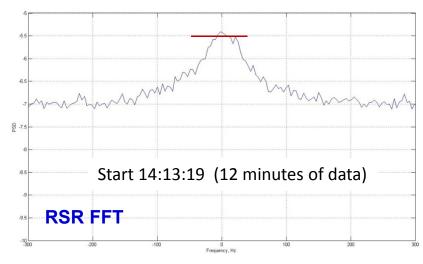
Jet Propulsion Laboratory
California Institute of Technology

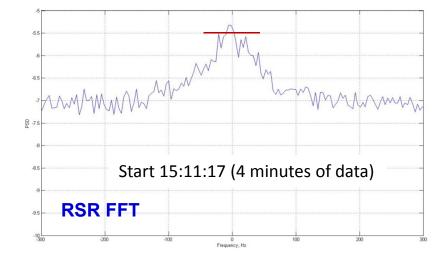


DSS-24/25 Phasing Experiment (DOY-121):

- DSS-24/25 transmitted 20 kW (carrier, PN11 code)
 - Array EIRP equivalent to 70m, transmitting 20 kW
- Successfully recorded DSS-24 carrier echo
 - Doppler-broadened carrier observed after 1 RTLT
 - RSR FFT: signal ~ 0.45 dB above noise-floor
- Successfully recorded DSS-24+DSS-25 carrier echo
 - RSR FFT: signal ~ 1.5 dB above noise-floor
 - Array combining gain: 3.7 (5.7 dB) over DSS-24
 - Array remained phased up throughout the track

DSS-24+DSS-25; signal peak about 1.5 dB above noise-floor

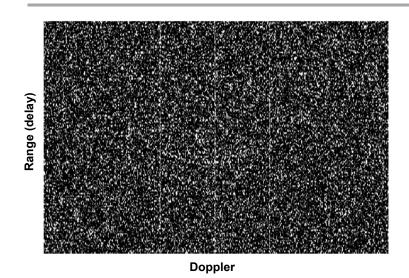




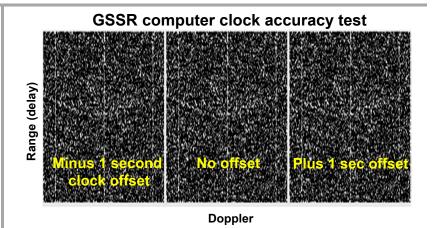


Mercury Imaging Experiment (DOY-121, 2010)

- Two 34m Apollo antennas, DSS-24/25, were phased up, transmitted 20 kW Doppler-compensated PN-modulated X-band carrier towards Mercury
- GSSR receiver at DSS-13 recorded faint Mercury signature



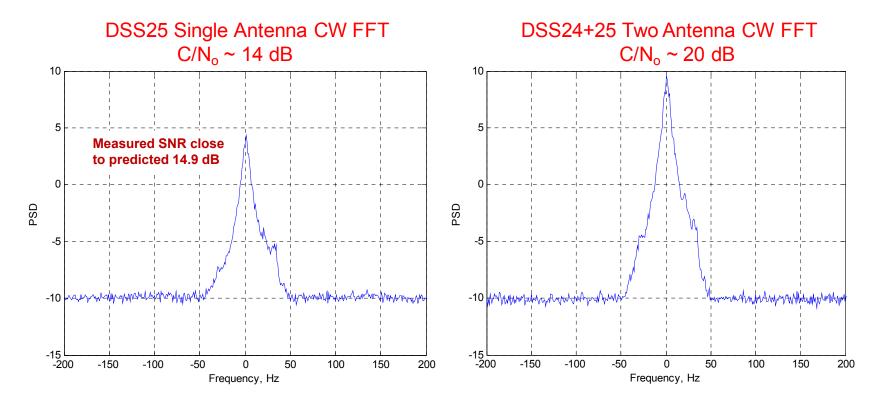
Original Mercury Doppler-delay image, obtained with GSSR receiver at DSS-13 on DOY-121, 1510-1550 UTC



Processed data to improve contrast, determine best computer clock setting, and extract image from data-files recorded after 1600 UTC with RSR and GSSR receivers

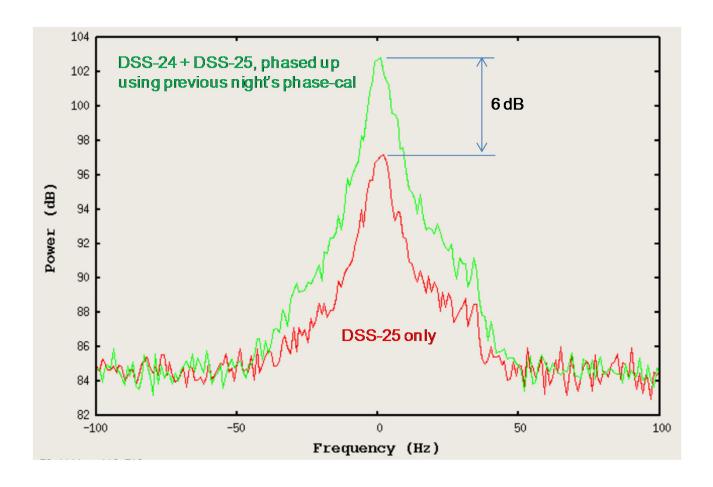


CW ECHOS FROM VENUS, DOY-294, RSR RECEIVER (DSS-13)



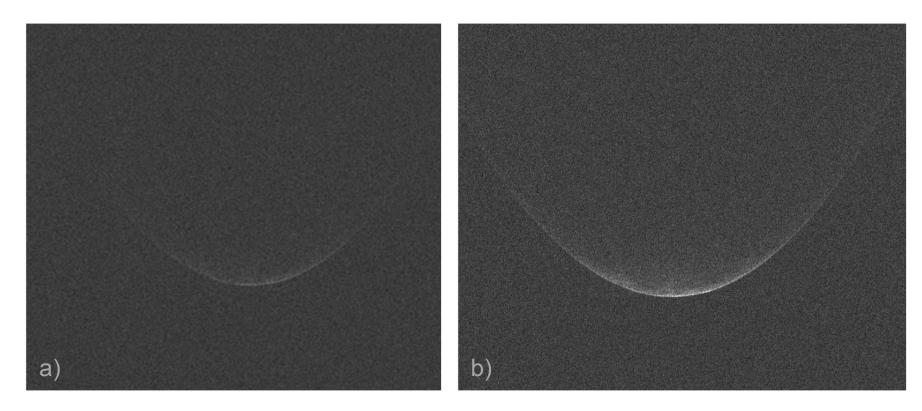
Venus echo for single and two-antenna Uplink Array illumination with RSR processing at DSS-13: SNR's of 14 and 20 dB as predicted, and demonstrating 6 dB array gain for the two-antenna array.





Venus echo for single antenna and two-antenna Uplink Array illumination with GSSR processing, demonstrating 6 dB array gain for the two-antenna array.

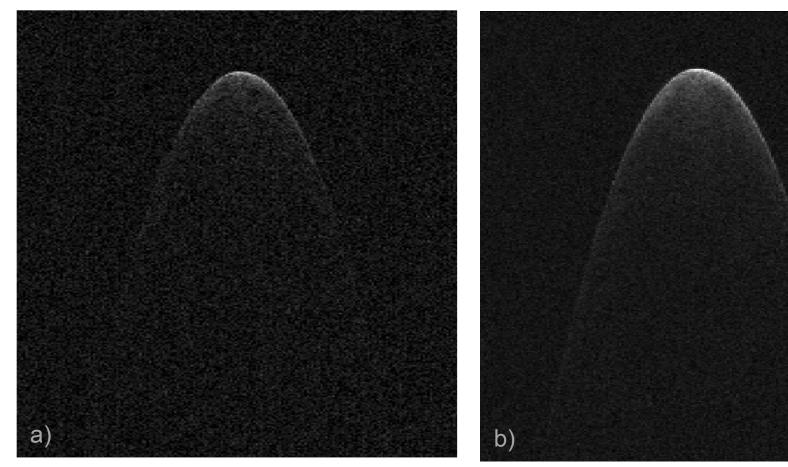




Doppler-delay images of Venus, taken on DOY-297, GSSR processing (4096 FFT): a) Single Antenna illumination (DSS-25); b) Two-antenna phased-array illumination (DSS-24 + DSS-25), showing greatly improved image quality.

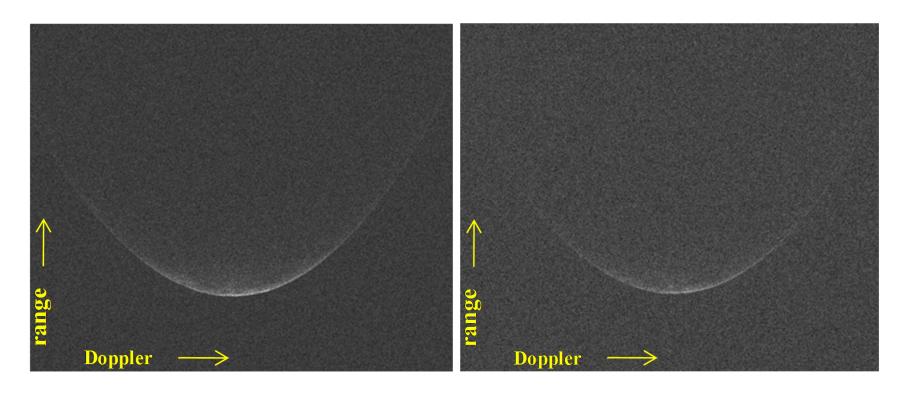
Range increases along the vertical axis, Doppler along the horizontal axis.





Doppler-delay images of Venus, taken on DOY-297, RSR processing (1024 FFT):
a) single antenna illumination (DSS-25); b) two-antenna phased-array illumination (DSS-24 + DSS-25), showing greatly improved image quality. Range decreases along the vertical axis, Doppler increases along the horizontal axis.

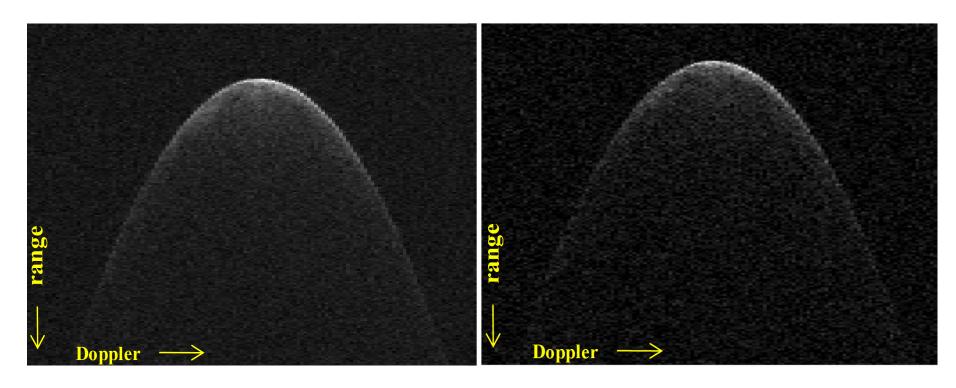




Attempted resolution of north-south ambiguity, GSSR processing (4096 FFT):

- a) +90 degree phase applied, northern hemisphere,
- b) -90 degree phase applied, southern hemisphere.
- c) Range increases along the vertical, Doppler along the horizontal axis.





Attempted resolution of north-south ambiguity, RSR processing (1024 FFT):

- a) +90 degree phase applied, northern hemisphere,
- b) -90 degree applied, southern hemisphere.

Range decreases along the vertical, Doppler increases along the horizontal axis.

Jet Propulsion Laboratory
California Institute of Technology

Reports, Conference publications, and NTRs to date

- 1. V. Vilnrotter, R. Mukai, D. Lee, "Uplink Array Calibration via Far-Field Power Maximization," IPN Progress Report 42-164, February 15, 2006.
- 2. NTR # 42597: "Uplink Array Calibration via Moon-Bounce Power Maximization" (Vilnrotter, Mukai, Lee)
- 3. JPL News Note: "JPL Performs First Two-Antenna Uplink Array Experiment" (appeared on 3/21/06)
- 4. NTR # 43674: "Uplink Array Calibration Using Power Measurements from a Nearby Spacecraft" (Vilnrotter, Lee, Paal, Mukai, Cornish)
- 5. F. Davarian, V. Vilnrotter, "Uplink Antenna Arraying for the Interplanetary Network," 24th AIAA International Communications Satellite Systems Conference, San Diego, CA, June 13, 2006.
- 7. V. Vilnrotter, Dennis Lee, "Uplink Array Experiment with the Mars Global Surveyor (MGS) Spacecraft," IPN-Progress Report 42-166, August 15, 2006.
- 8. L. Paal, R. Mukai, T. Cornish, V. Vilnrotter, D. Lee, "Measurement of Antenna Phases due to Ground Equipment Effects in an Uplink Array," to appear in IPN-Progress Report 42-167, November 15, 2006.
- 9. L. Paal, R. Mukai, V. Vilnrotter, T. Cornish, and D. Lee, "Ground System Phase Estimation Techniques for Uplink Array Applications," IPN Progress Report 42-167, November 15, 2006.
- 10. V. Vilnrotter, D. Lee, R. Mukai, T. Cornish, P. Tsao, "Three-antenna Doppler-delay Imaging Of The Crater Tycho For Uplink Array Calibration Applications," IPN Progress Report 42-169, May 15, 2007.
- 11. Doppler-Delay Calibration of Uplink Arrays via Far-Field "Moon-Bounce" Power Maximization, V. Vilnrotter, D. Lee, R. Mukai, T. Cornish, P. Tsao, 11-th ISCOPS Conference, Beijing, May 15, 2007.
- 12. NTR # 44611: "Ground System Phase Estimation for Uplink Arrays," L. Paal, R. Mukai, V. Vilnrotter, T. Cornish, D. Lee, November 1, 2006.

Jet Propulsion Laboratory
California Institute of Technology

Reports, continued

- 12. NTR # 45243: "A Program for Calculating Pointing, Doppler, and delay compensation for the Moon-Bounce Experiment," V. Jamnejad, V. Vilnrotter, N. Bachman, August 2, 2007.
- 13. V. Vilnrotter, P. Tsao, D. Lee, T. Cornish, V. Jamnejad, "Results of EPOXI Uplink Array Experiment of June 27th, 2007," IPN Progress Report 42-174, August 15, 2008.
- 14. V. Vilnrotter, D. Lee, T. Cornish, P. Tsao, L. Paal, V. Jamnejad, "Uplink Array Concept Demonstration with the EPOXI Spacecraft," IEEE Aerospace Conference, Big Sky, MO, March 12, 2009.
- 15. P. Tsao, V. Vilnrotter, V. Jamnejad, "Pointing-Vector and Velocity Based Frequency Predicts for Deep-Space Uplink Array Applications," IEEE Aerospace Conference, Big Sky, MO, March 12, 2009.
- 16. V. Vilnrotter, P. Tsao, D. Lee, T. Cornish, L. Paal, "Uplink Array Calibration via Lunar Doppler-Delay Imaging," IEEE Aerospace Conference, Big Sky, MO, March 5-12, 2010.
- 17. V. Vilnrotter, K. Andrews, J. Hamkins, A. Tkacenko, "Maximum Likelihood Estimation of Navigation Parameters from Downlink Telemetry," IEEE Aerospace Conference, Big Sky, MO, March 5-12, 2010.
- 18. V. Vilnrotter, K. Andrews, A. Tkacenko, J. Hamkins, "Maximum Likelihood Estimation of Navigation Parameters from Downlink Telemetry," IEEE Aerospace Conference, Big Sky, MO, March 5-12, 2010." 12th ISCOPS Conference, Montreal, Canada, July 27, 2010.
- 19. V. Vilnrotter, P. Tsao, D. Lee, T. Cornish, J. Jao, M. Slade, "Planetary Radar Imaging with the Deep-Space Network's 34 meter Uplink Array," IEEE Aerospace Conference, Big Sky, MO, March 5-12, 2011.